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20. ABSTRACT (Continue on reverse side if necessary and identify by block number)

The achievement of laser fusion requires the inward acceleration of pellet shells to very high velocity without excessive preheat of the pellet fuel. In a study of ablatively accelerated thin planar Al foils, we have determined the complete time history of the rear surface temperatures, and evaluated the energy transport mechanisms that contribute to the heating. Extrapolation to larger, reactor-sized pellets indicates that preheat is minimal at low laser irradiances.

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PREHEAT STUDIES ON LASER ABLATIVELY-ACCELERATED FOILS

In current laser fusion studies, it is recognized that preheat

of the target fuel of more than a few electron volts can significantly

reduce the degree of compression and pellet yield that can be achieved.[1]

Therefore, it is of considerable importance to know the temperature on

the inside of the target shell. Such knowledge will allow one to choose

the proper isentrope for compression calculations as well as to check the

predictions of detailed numerical simulations of laser driven implosions.

Using laser-driven ablation we have accelerated thin planar foils, which

simulate the early behavior of an imploding pellet shell, to velocities

of 1.5×10^7 cm/sec for CH targets and 3×10^6 cm/sec for Al targets.[2]

Here we report measurements of rear surface temperatures of accelerated Al

targets which remain below 3 eV through the bulk of the laser pulse, achieving

a directed kinetic energy per particle greater than 80 times the temperature.

The temperatures are inferred by measuring the absolute rear surface

continuum emission with monochromators and then using blackbody emission

assumptions, which we verify.[3] These are the first time-resolved temperature

measurements for these conditions. Similar techniques have been used

elsewhere to measure temperatures and velocities associated with high-

pressure shock phenomena.[4,5]

The experiments were performed using one beam of the NRL Pharos II

Nd:phosphate-glass laser system (1.05 μm) with up to 200 joules on target

in a 4-nsec FWHM pulse. Irradiation uniformity is improved and edge

effects are minimized by using large focal spot diameters (typically 1 mm)

obtained by placing the target off-focus toward the laser to produce

irradiances of $1-6 \times 10^{12}$ W/cm^2. Temperature measurements were made on

Al foil targets ranging in thickness from 4-23 μm. The experimental

setup is shown in Fig. 1. Two 3/4-m-monochromators equipped with 1-nsec rise time photomultipliers[6] measure continuum intensity from the rear surface of the target at two wavelengths simultaneously. Using a beam splitter, both monochromators make observations along a common axis positioned 6 degrees off the laser axis (but normal to the target axis). Good shot-to-shot reproducibility allows a parameter study in target thickness, target material, and wavelength of observation. The demagnified image of the spectrograph slit (see Fig. 1, inset), samples a central portion of the focal spot area. Typically, the slit widths are set to observe a 17-33 $\overset{o}{A}$ bandwidth. An absolute calibration of the optics, the spectrographs, and the photomultipliers is made "in situ" using a tungsten ribbon-filament lamp, calibrated by the National Bureau of Standards.[7]

A survey over the spectral range 3500-6500 $\overset{o}{A}$ was made by taking time-integrated spectra both parallel and perpendicular to the rear target surface. The continuum wavelength intervals were chosen at positions where no spectral lines were observed. Several experiments were done to check the assumption that the rear surface radiates as a blackbody. Figure 2 shows peak intensity versus wavelength data for Al foils of 4 μm, 7 μm, and 12 μm thickness. This figure demonstrates the magnitude of shot-to-shot scatter and calibration errors. It is seen that the intensity data have the magnitude and approximate wavelength dependence expected from a blackbody.

In addition, the peak intensity of a strong aluminum spectral line was found to be the same ($\pm20\%$) as the neighboring continuum. This would be the case if the radiation was coming from an isothermal, optically-thick plasma, i.e., a blackbody. We conclude that the blackbody assumptions are justified.

LASER
BEAM
$\lambda = 1.05 \mu m$

6°

TARGET
CHAMBER

LENS

6.3°

|—1 mm—|

LASER
FOCAL
SPOT

IR BLOCKING
FILTER

SLIT IMAGE

BEAM SPLITTER
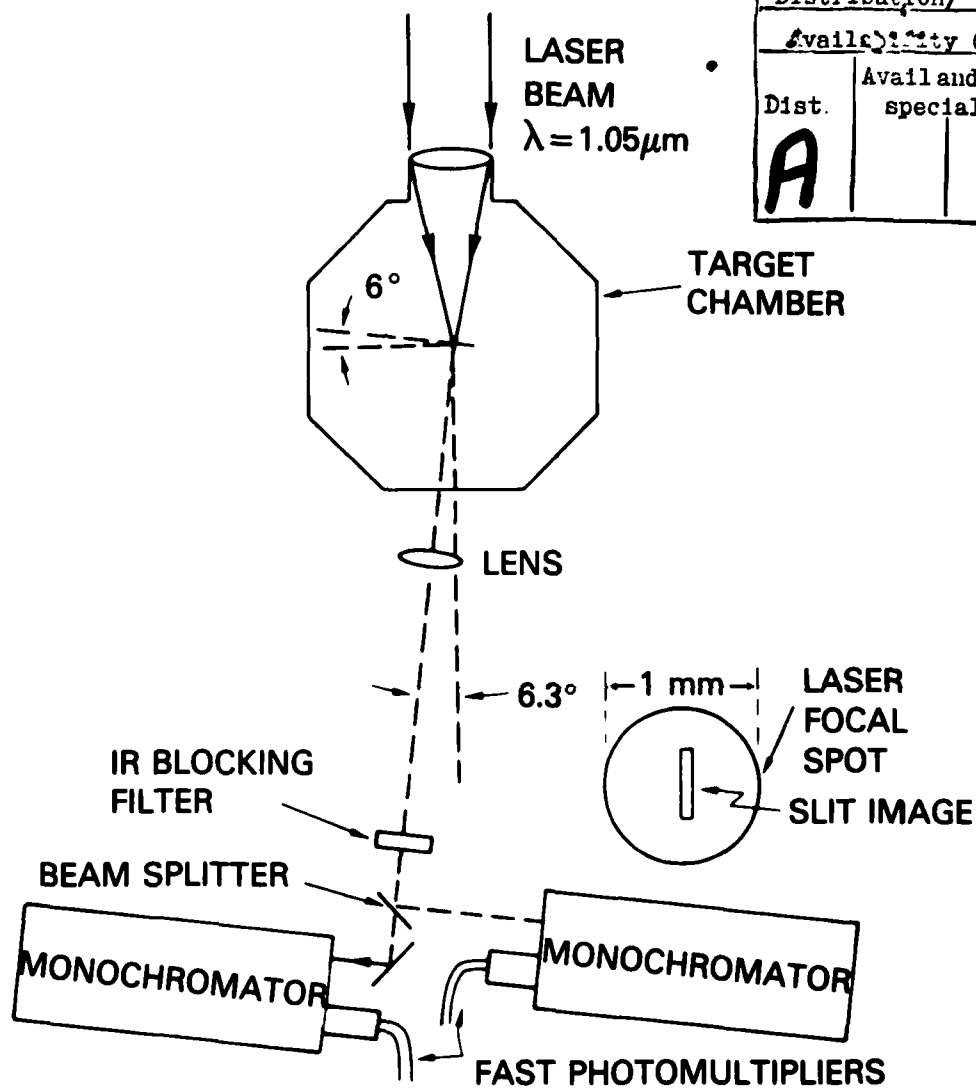
MONOCHROMATOR

MONOCHROMATOR

FAST PHOTOMULTIPLIERS

Fig. 1 — Experimental setup showing collection optics and field of view of monochromators.
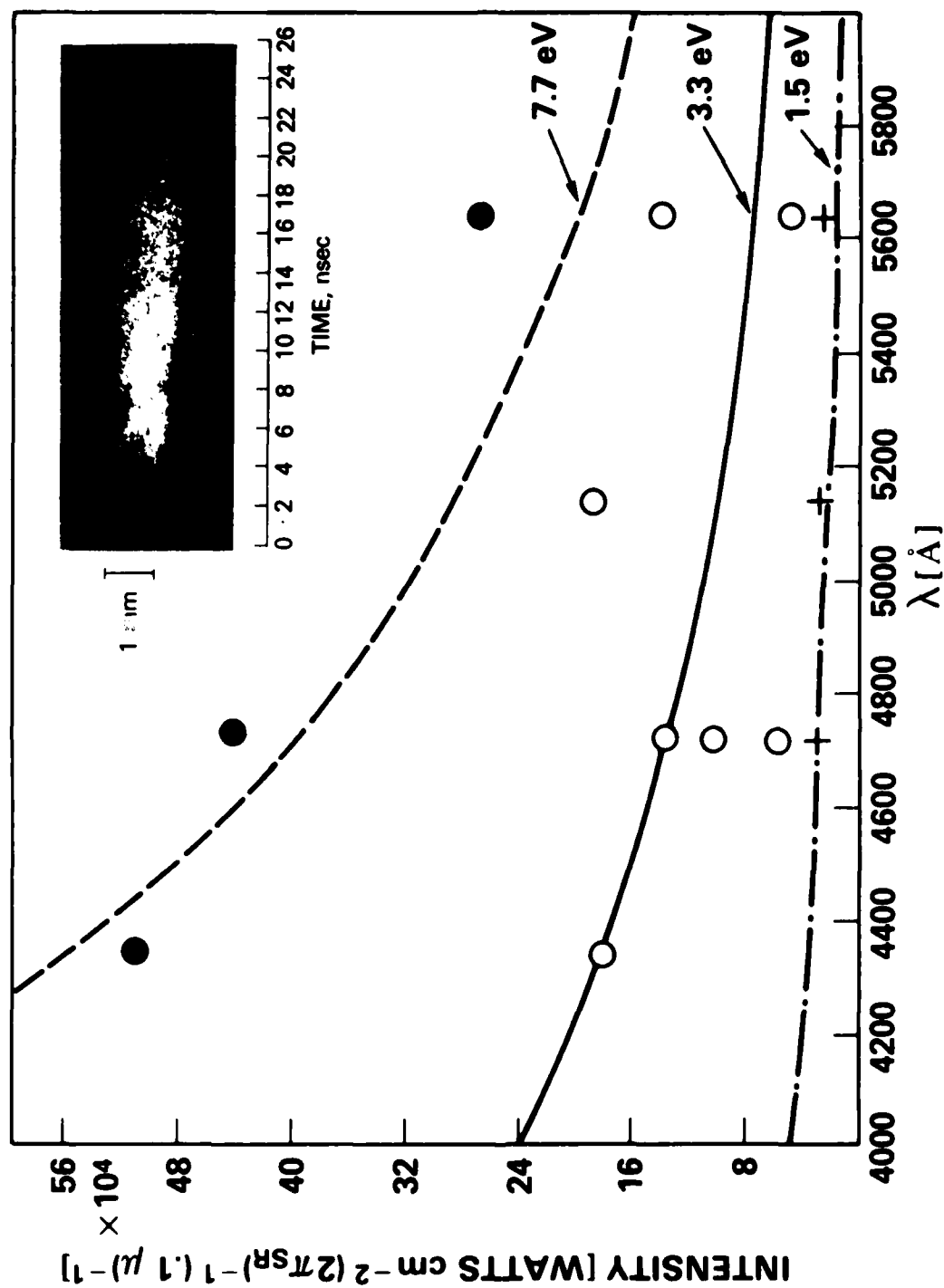
Fig. 2 — Rear surface continuum intensity vs. wavelength for 3 Al foil thicknesses at 5×10^{12} W/cm^2. Curves are black-body intensities at the given temperatures. (Data points: $\bullet$ = 4μm Al; o = 7μm Al; + = 12μm Al). Inset shows streak photograph of rear surface luminosity across a diameter of the focal spot.

Interferometric plasma density measurements on the rear surface of the accelerated target obtained with a laser (5270 $\overset{o}{A}$) probe beam passing tangentially to the target surface gave maximum densities of approximately 10^{19} cm^{-3} and a gradient scalelength of 50 μm at the visible rear surface. However, these density measurements are limited by the opacity of the 1 mm plasma path length for the probe light, and by refraction of the probe light out of the f/2 collection optics due to a steep electron density gradient. Opacity calculations[8] show that, at the emission wavelengths observed and at a temperature of 0.5-2 eV, the plasma has an opacity of about 1 for a density of several times 10^{20} cm^{-3} and a path length of 10 microns. Therefore, the temperature we are measuring characterizes a region in the density interval 5 x 10^{19} cm^{-3}< n <5 x 10^{20} cm^{-3}. (An optical probe beam directed perpendicularly at the rear side of the target was found to be strongly absorbed when the temperature reached a few tenths of an electron volt.)

Figure 3 shows the time history of the rear surface temperatures for Al foil targets of several different thicknesses. Accuracy of the temperature determination is better than a factor of 2 for T > 2 eV and improves to ±20% at T_e = 0.4 eV. In each case, heating occurs during the period of the laser pulse but the temperature continues to increase for several nanoseconds thereafter. The peak temperatures are seen to shift to later times and lower values as the target thickness increases. Figure 3(inset) shows the dependence of temperature on target thickness just after the acceleration (which is over by about +2 to+3 nsec). Additionally, the peak rear surface temperature was found to increase linearly with increasing irradiance (or energy) over the irradiance range 2 - 6 x 10^{12} W/cm^2. The peak temperature (T_e) for Al as well as CH and C foils, all fit a single

hyperbola, $T_e \, \rho_A = 8 \times 10^{-3}$ eV gm cm^{-2} at 5×10^{12} W/cm^2, where the

areal density ρ_A is the product of material density and target thickness.
The peak temperatures ranged from 34 eV for a 2.5 μm CH foil to 0.8 eV for
23 μm Al foil.

Possible mechanisms for energy transport to the rear of targets
include fast electrons, x rays, thermal conduction and shocks. The
contributions of these phenomena have been evaluated by separate
measurements and by considerations of the time history and parametric
behavior of the measured temperature.

First, the lack of measurable x-ray emission (0.1 erg) in the energy
range 20-50 keV, (corresponding to radiation from electrons with ranges com-
parable to target thickness) from the front side of the foils[9] demonstrates
that there is insufficient fast electron energy (5×10^{-5} J) to cause appre-
ciable heating of the rear target surface.

Second, time-resolved studies have shown that the x-ray emission
(>1 keV) very closely follows the time history of the incident laser irradiance,
so that x-ray preheat effects should closely follow the time-integrated laser
pulse (Fig. 3). Based on the known range and measured intensities of x rays
in the energy range of 1.5 - 50 keV emitted from the front side of Al tar-
gets, the important x rays for the purpose of preheating the rear target
surface are the x-ray line and continuum radiation between 1.6 and 2.3 keV.
(The much lower intensity continuum emission just below the K-edge at 1.56 keV,
as yet unresolved, may also be an important preheat agent.) The time-integrated
x-ray line spectrum was measured on an absolute basis (to an estimated
accuracy of a factor of 2) and the x-ray energy deposited throughout the
target was calculated, assuming spatial uniformity of the x-ray source
and using cold material absorption coefficients. A knowledge of the equation

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of state is necessary in order to convert preheat energy deposition to temperature. We assume the target foil material remains near solid density for the times of interest, and use tabulated heat capacity data[10] to determine the temperature resulting from the calculated local x-ray deposition (curve at 3 nsec, Fig. 4). We find that the rear-side temperature rise during the laser pulse is consistent with 1.6 - 2.3 keV x-ray deposition.

Third, using formulae and tabulated data for thermal conduction and radiative diffusion[10] and the tabulated data for heat capacities, numerical calculations assuming a fixed temperature at a point 1 μm in from the front surface and no losses from the rear surface (curves at 6 and 12 nsec, Fig. 4) show that thermal conduction may explain the upward drift in temperature after termination of the laser pulse.

Fourth, the contribution to the heating of the target due to shock waves that may be present is difficult to evaluate precisely. From measurement of ablation particle velocities, as well as gross acceleration of the target foil, we infer a peak ablation pressure of $\sim$2 Mbar at 5×10^{12} W/cm^2 on Al. A single 2 Mbar shock propagating into cold Al will produce a 75% density increase, a 0.8 eV temperature rise, and a shock velocity of 1.3×10^6 cm/sec.[10] In fact, the small early temperature rise for the thicker Al targets (12-23 μm) appears consistent with such a model. For thin targets, however, one must use a hydrodynamic code to evaluate the magnitude of shock wave heating; multiple reflections occur because the rise time of the pressure pulse is longer than the shock transit time through the target. Two different hydrodynamic codes, which have been run for our experimental conditions, indicate that the x-ray deposition is the major contributor to the rear surface heating for 4-12 μm Al targets.[11,12] The codes agree within a factor of two with the observed temperature and closely match the temporal behavior for these targets.
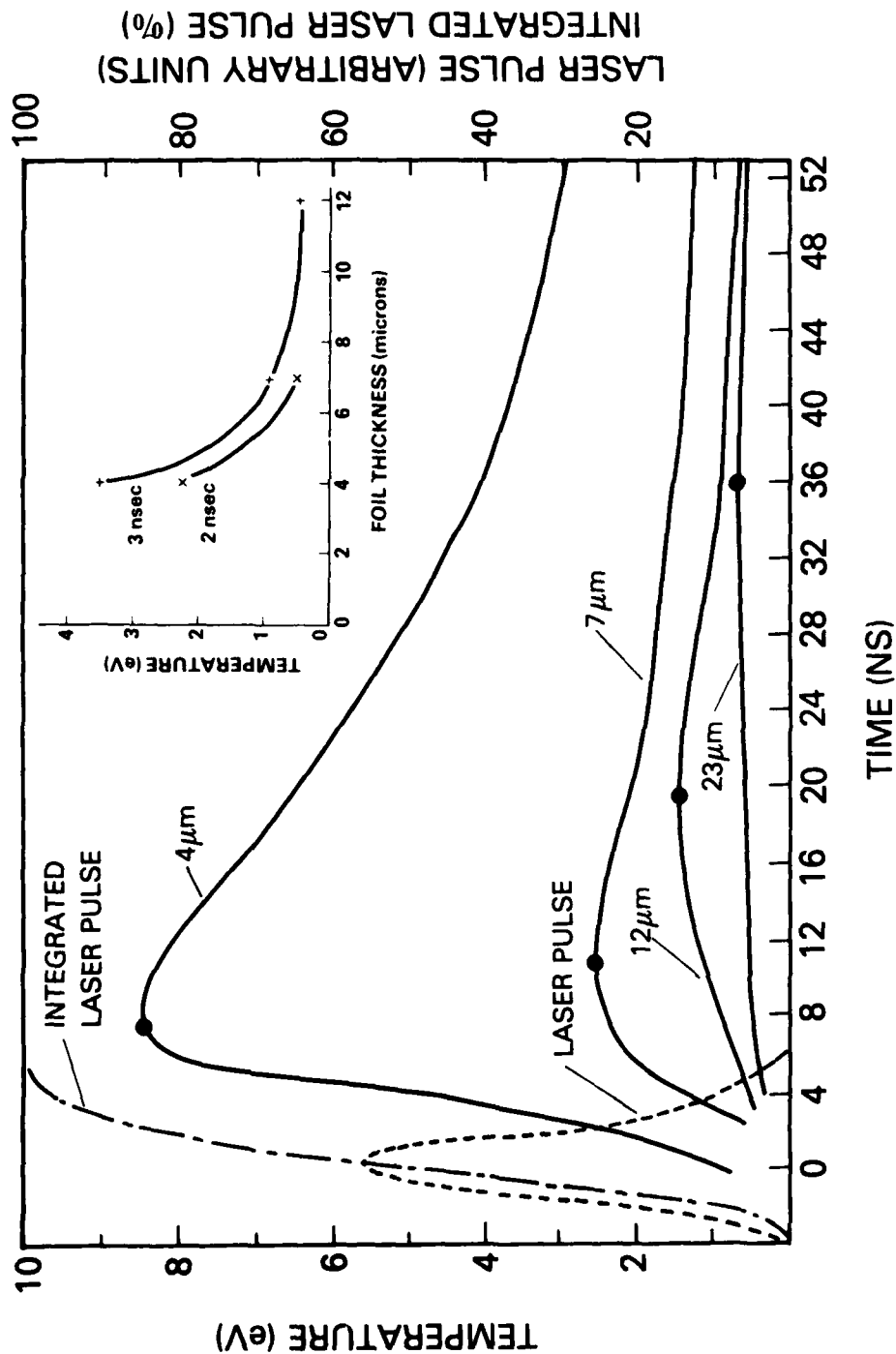
Fig. 3 — Rear surface temperature vs. time for 4 Al foil thicknesses at 5×10^{12} W/cm², 1 mm spot, $\lambda = 4717$ Å, $\Delta\lambda = 33$Å. Dotted curve is the time history of the laser pulse and the dashed curve is the integrated laser pulse. The time t = 0 corresponds to the time of the peak of the laser pulse. Inset shows rear surface temperature vs. Al foil thickness at two times at the end of the acceleration process.
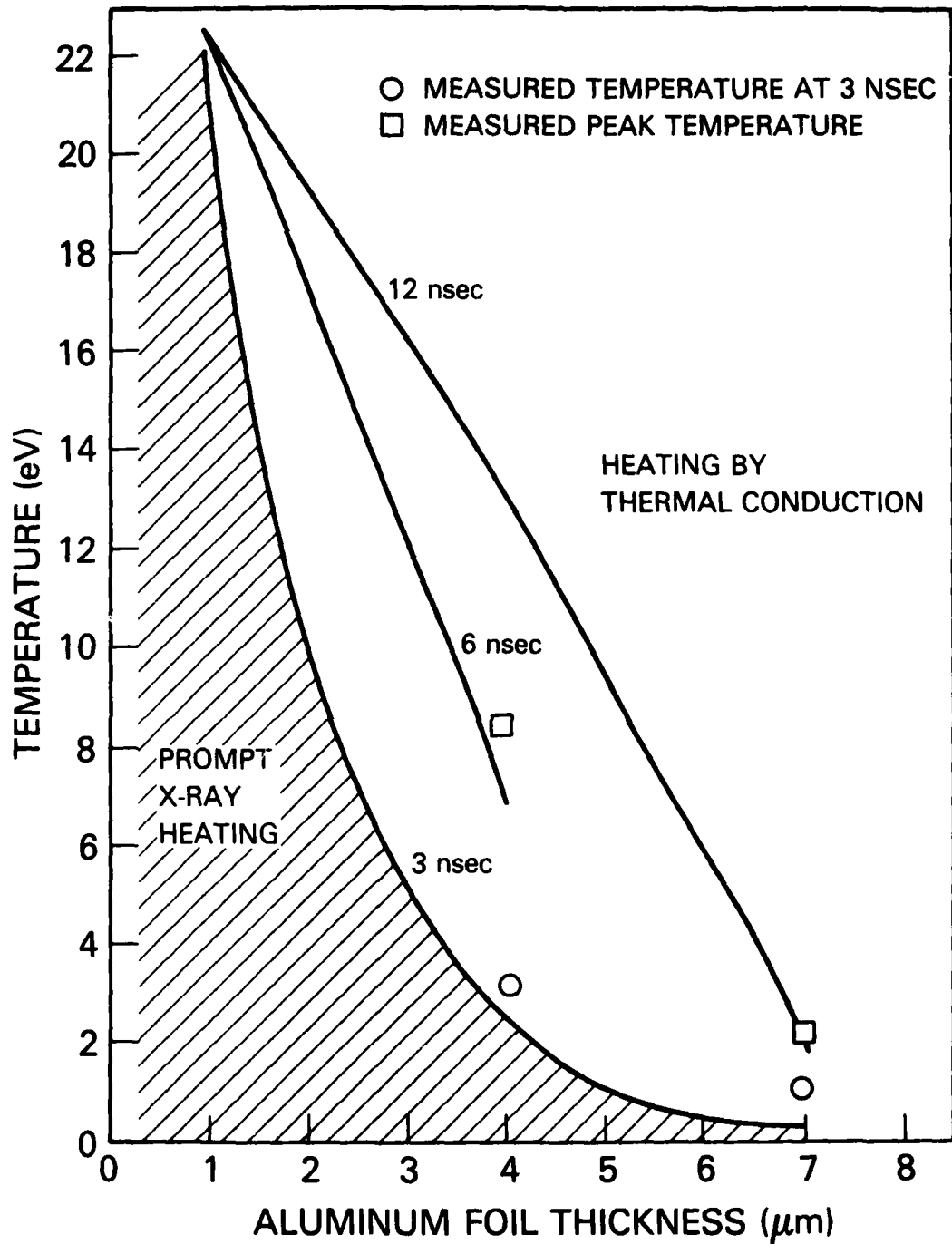
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Fig. 4 — Rear surface temperature vs. Al foil thickness showing the prompt x-ray heating at 3 nsec, and prompt plus thermal conduction heating at 6 nsec for the 4 μm foil and at 12 nsec for the 7 μm foil. The square data points are measured temperature values at 6 nsec for the 4 μm Al foil and at 12 nsec for the 7 μm Al foil.

The late time heating and cooling of the rear surface can be understood in terms of an energy transport balance into and out of the rear surface. Thermal conduction calculations similar to those previously mentioned suggest that ordinary axial thermal conduction can account for the slow upward drift of the rear surface temperature that is seen on the thicker targets. The importance of lateral thermal conductivity on the late time development of the rear surface temperature is indicated by measurements on a 7- μm-thick Al foil. These show that when the focal spot size is reduced from 1 mm to ¼ mm diameter at constant irradiance, the peak temperature is lower by 50% and occurs earlier in time, and the temperature decays more rapidly.

We have examined the possibility of energy flowing around our target foil, which has been reported on disc targets with higher irradiance CO_2 lasers.[13] By increasing the width of the foil from 3 mm to 1 cm, we see less than 10% change in the rear surface temperature, and conclude that the effect of energy flowing around the target, if present, is small. There is also the possibility of energy flowing around the edges of the slug of accelerated material ("cookie-cutter" model).[14] Streak photography of the rear surface visible emission (Fig. 2, inset), as well as time-integrated photography, show no sign of either of these effects.

In conclusion, we have measured the time history of the rear surface temperature of foils accelerated by laser ablation. The relative effects of x rays, shocks, thermal conduction, and fast electrons upon preheating this surface have been evaluated. It is concluded that for our conditions direct heating by soft x rays followed by thermal conduction appears capable of explaining most of the heating observed in thinner foils. Shock waves

are more important for the thicker targets. The energy flows through rather than around the targets, providing a close analogy to the laser ablative implosion of pellet shells.

It should be noted that in these experiments the rear surface temperature of even the thinnest Al targets (4 μm) remain on a low isentrope ($\lesssim 3$ eV) throughout the acceleration process. Proposed low-irradiance, reactor-sized pellets[15] would have thicker shells ($\gtrsim 50$ μm) and thus even less heating by x rays and thermal conduction.

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